

PNEUMATIC

Test stand for air turbine starters

>TATS2EF<



The test stand is designed to test Air Turbine Starters for the aircraft types Eurofighter "EF2000" and "F18-Hornet". It is developed for tests in combination with the Air Turbine Starter Motor Control Valve.

It is possible to adapt this equipment for other aircraft types

- Tests can be carried out manually, semi-automatic or fully automatic
- > An electric motor is used to simulate the starter moment of inertia, and returns generated current into the power system
- > Gearbox:

One shaft with a max. of 70 000 rpm One shaft with speeds of 3 000 / 10 000 rpm (hydraulically switched)

- > Case drain testing is carried out
- > All required parameters e.g. pressure, flow, temperature, torque, rpm, vibration, time and resistance are measured, indicated and stored electronically

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MISCELLANEOUS

- > Two lubricating oil units are provided. One for the UUT and one for the test stand
- > The temperature can be regulated during lubrication of the UUT
- > A separate air supply is provided
- > A separate control console for use outside the test chamber is also supplied
- > A modem is fitted to the test stand to allow the software to be maintained directly at the factory
- > Calibration is carried out autonomously by the software

TECHNICAL DATA

> Pneumatic parameters:

Compressed air supply for the starter inlet (dynamic):

Pressure 0-7 bar (0-102 psi) Temperature up to 250°C (482°F)

Flow max. 1 kg/s (max. 132 ppm)

Connector 4"

Outflow air for the starter outlet:

Pressure ambient

Temperature max. 250°C (482°F)

Flow max. 1 kg/s (max. 132 ppm)

Connector 5"

Leakage test circuit for the housing (static):

Pressure 0-2 bar (0-29 psi)

Temperature ambient

Hydraulic parameters (Lubricating oil unit for the UUT):

Hydraulic power unit:

max. 5 bar / max. 4.0 lpm (max. 73 psi / max. 1.1 USgpm) Oil temperature: up to 100°C (212°F) Capacity reservoir: 30 l (7.9 USgal)

> Hydraulic parameters (Lubricating oil unit for the gearbox):

Hydraulic power unit:

max. 6 bar / max. 20.0 lpm (max. 87 psi / max. 5.3 USgpm) Oil temperature: max. 45°C (113°F) Capacity reservoir: 80 l (21.1 USgal)

Medium (for the lubricating oil units):

MIL-PRF 23699F Grade STD

> Electrical supply:

3/N/PE AC 50 Hz 400 V (test stand)

Nominal current: 220 A Power: 140 kVA Fuse: 250 A

1/N/PE AC 50 Hz 230 V (control panel)

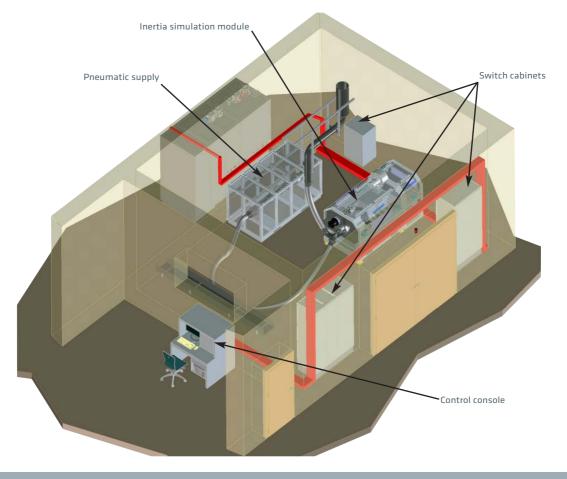
Nominal current: 13 A Fuse: 16 A

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TECHNICAL DATA

> Measurement range:			> Dimensions:		
Pressure:	0-7 bar (0-102 psi)	Cl. 0.25	Inertia simulation module		
	0-2.5 bar (0-36 psi)	CI. 1	Length:	2450 mm	(8.0 ft)
	0-6 bar (0-87 psi)	Cl. 0.25	Width:	1150 mm	(3.8 ft)
			Height:	1800 mm	(5.9 ft)
Differential					
pressure:	0-200 mbar	Cl. 0.25	Control console		
			Length:	1300 mm	(4.3 ft)
Flow:	0-1 kg/s (0-132 ppm)	Cl. 2	Width:	950 mm	(3.1 ft)
			Height:	1370 mm	(4.5 ft)
Temperature:	0-300°C (572°F)	± 2°C (36°F)			
			Pneumatic supply		
Vibration:	0-5 g	± 7% o.m.r.	Length:	2200 mm	(7.2 ft)
			Width:	1130 mm	(3.7 ft)
Speed:	0-12000 rpm	± 2 rpm	Height:	1250 mm	(4.1 ft)
Torque:	0-500 Nm	CI. 1			
Resistance:	0-600 Ohm	±10hm			

TYPICAL INSTALLATION OF THE FACILITY



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Measuring cabinet

OPTIONS

Many options are possible for adaption, e.g.adaption to other aircraft types, to different touch-screens etc.

Technical data are subject to change!